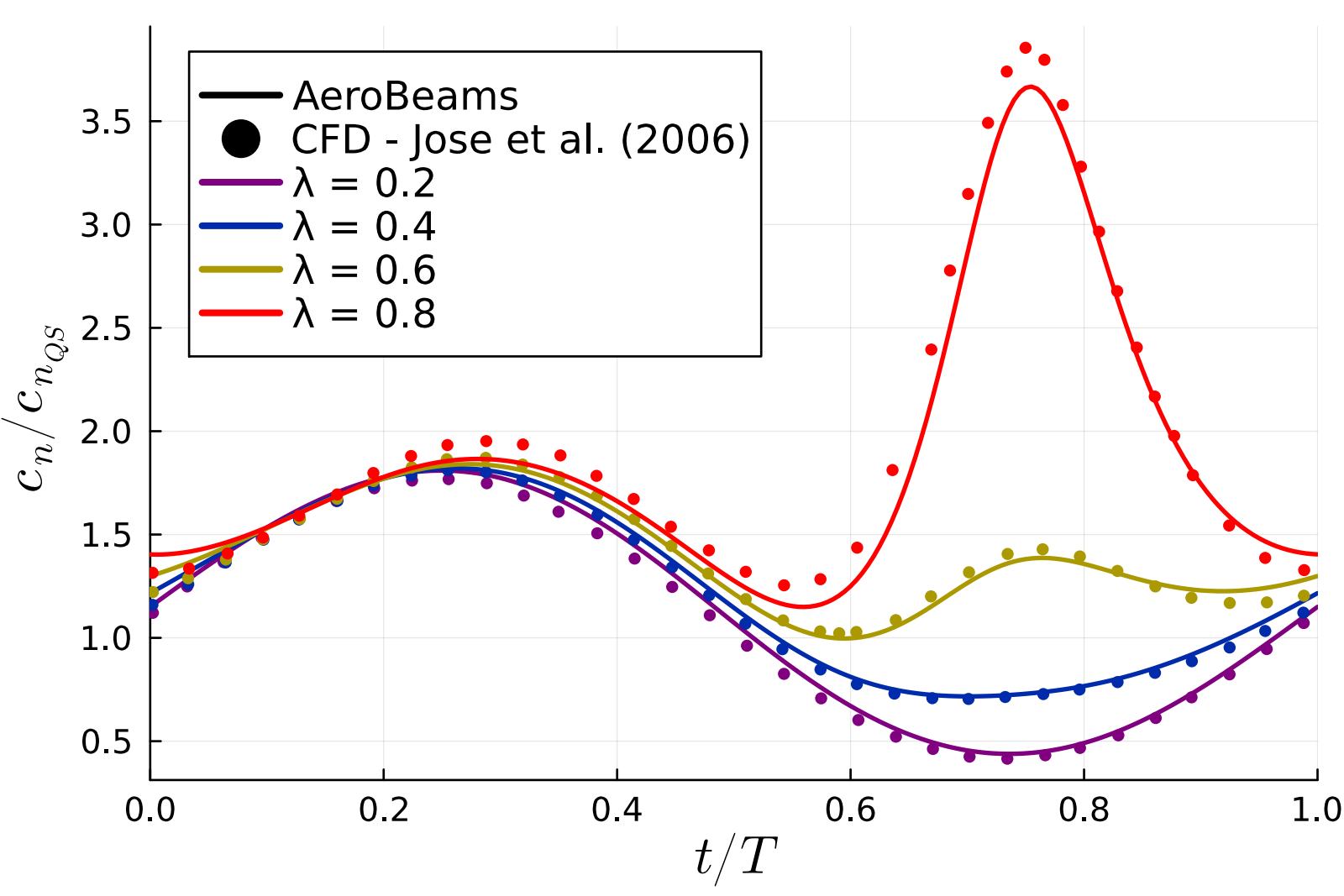

AeroBeams
CFD - Jose et al. (2006)
λ = 0.2
λ = 0.4
λ = 0.6
λ = 0.8
$c_n/c_{n_{QS}}$
t/T
0.5
1.0
1.5
2.0
2.5
3.0
3.5
0.0
0.2
0.4
0.6
0.8
1.0